

SUPPORTING INFORMATION

for:

Physical characterization of the fine particle emissions from commercial aircraft engines during the Aircraft Particle Emissions eXperiment (APEX) 1 to 3

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This document provides supplemental information and data as cited in the text of the published manuscript.

1. Sampling Probe

During APEX-1, -2, and -3, a single sampling probe located at the centerline of the engine exit plane was used at a distance of 15-, 30-, or 43-m behind the engine. Fig. S-1 shows a photo of the sampling probe used in all three APEX sampling campaigns at the 30-m location.



Fig. S-1. Photo of APEX-1 30-m sampling probe.

2. Measurement Parameters

Both continuous monitoring and time-integrated sampling was conducted during the three APEX sampling campaigns. A complete list of all parameters measured along with the sampling location, type of sample, and specific instruments used is shown in Table S-1 as installed in the Diesel Emissions Aerosol Laboratory (DEAL). The DEAL was used as the sampling platform for all APEX emissions sampling.

3. Test Engines

A total of nine different engine models were tested during the APEX program. Specifications for each engine tested are shown in Table S-2. Note in this table that different models of the same engine family are identified by the alpha-numeric characters behind the dash mark (e.g., the model number is 7B24 for the CFM56-7B24). Table S-2 also provides smoke number data as obtained during new engine certification per International Civil Aviation Organization (ICAO) requirements. Although smoke number was intended to reduce visible emissions, it is also a relative indicator of the PM emissions from the engine.

4. Experimental Matrix

A total of 24 tests were conducted by EPA during the three APEX sampling campaigns. Table S-3 provides the test number, aircraft and engine type, power conditions, and fuel flows evaluated during each test. It should be noted that the fuel flow rate values shown in the table are averages of all the data obtained under the same power (percent rated thrust) condition occurring during a test. In some cases, acceptable data may not be available for all periods at a particular thrust level and thus these values may not match those shown in Fig. 2 of the main paper. Fig. 2 of the main text shows the average fuel flows obtained only from those periods during a particular test that had valid nano-SMPS measurements.

Table S-1.

Measurements performed by the DEAL during APEX-1, -2, and -3.

Parameter	Sampling Location	Measurement Technique	Type of Sample	Instruments and Sampling Media
PM-2.5 mass concentration	Background	Microbalance	Continuous	Rupprecht and Patashnick (now Thermo Electron) Series 1400a TEOM
	Background	Gravimetric analysis	Time-integrated	47-mm Teflon filter with double quartz backup filters for collection of gas-phase “blow off” ^a
	Plume	Microbalance ^b	Continuous	Rupprecht and Patashnick Series 1105a TEOM
	Plume	APEX-2 & -3: QCM ^b	Continuous	SEMTECH Model RPM-100 particulate monitor + diluter
	Plume	Gravimetric analysis	Time-integrated	47-mm Teflon filter with double quartz backup filters ^a
Particle size distribution	Background	Low pressure cascade impactor (aerodynamic diameter)	Continuous / time-integrated	Dekati ELPI
	Background	Electrical mobility classifier/condensation nuclei counter (electrical mobility diameter)	Continuous	APEX-1: TSI Model 3934 SMPS, Model 3071 A classifier, Model 3010 CPC APEX-2 & -3: TSI Model 3936 SMPS (long), Model 3080 classifier, Model 3025a CPC, Model 3081 DMA
	Plume	Low-pressure cascade impactor (aerodynamic diameter)	Continuous / time-integrated	Dekati ELPI
	Plume	Electrical mobility classifier/condensation nuclei counter (electrical mobility diameter)	Continuous	TSI Model 3936 SMPS (Nano), Model 3080 classifier, Model 3025a CPC, Model 3085 DMA
	Plume	APEX-1: electrical mobility classifier/condensation nuclei counter (electrical mobility diameter) APEX-2 & -3: electrical mobility classifier/electrometers (electrical mobility diameter)	Continuous	APEX-1: TSI Model 3936 SMPS (long), Model 3080 classifier, Model 3025 CPC, Model 3081 DMA APEX-2 & -3: TSI Model 3090 EEPS + diluter
PM-2.5 number concentration ^c	Background	Condensation nuclei counter	Continuous	Model 3025a CPC
	Plume	Condensation nuclei counter	Continuous	Model 3025a CPC + diluter
Elemental carbon/organic carbon (EC/OC)	Background	Thermo-optical analysis (NIOSH Method 5040)	Time-integrated	Prefired 47 mm quartz filter
	Plume	Thermo-optical analysis (NIOSH Method 5040)	Time-integrated	Prefired 47 mm quartz filter
	Plume	Optical attenuation/UV absorption (black carbon)	Continuous	TSI 3302a Diluter + Magee Model AE-2 Aethalometer ^d
PM semivolatile organic compounds (SVOCs)	Background	GC/MS	Time-integrated	Prefired 47 mm quartz filter with 4 backup PUF plugs. ^a
	Background	Low-pressure cascade impactor	Time-integrated	12 aluminum foil ELPI stages + prefired quartz back-up filter ^b
	Plume	GC/MS	Time-integrated	Prefired 47 mm quartz filter with 4 backup PUF plugs. ^a
	Plume	Low-pressure cascade impactor	Time-integrated	12 aluminum foil ELPI stages + prefired quartz back-up filter ^b
	Plume	UV analyzer (particle surface PAHs)	Continuous	EcoChem Model PAS 2000
Total volatile PM + EC/OC	Plume	Gravimetric/thermo-optical analysis	Time-integrated	Dekati Model EKA-111 thermal denuder with parallel Teflon and double prefired quartz filters
PM inorganic water-soluble ions	Background	Ion chromatography	Time-integrated	Teflon filter
	Plume	Ion chromatography	Time-integrated	Teflon filter

Parameter	Sampling Location	Measurement Technique	Type of Sample	Instruments and Sampling Media
PM elemental composition	Background	XRF	Time-integrated	Teflon filter
	Plume	XRF	Time-integrated	Teflon filter
APEX-1 CO, CO ₂ , total VOCs	Background	IR absorption	Integrated bag ^e	Brüel & Kjær Model 1302 Photoacoustic Analyzer
	Plume	IR absorption	Continuous	Brüel & Kjær Model 1302 Photoacoustic Analyzer
APEX-2 & -3 CO ₂	Background	IR absorption	Continuous	Milton-Roy (CA Analytical) Model 5300A
	Plume	IR absorption	Continuous	Milton-Roy (CA Analytical) Model 5300A
Gas-phase NMOCs	Background	GC/MS/FID	Time-integrated	SUMMA-passivated canister
	Plume	GC/MS/FID	Time-integrated	SUMMA-passivated canister
Gas-phase carbonyl compounds	Background	HPLC	Time-integrated	DNPH impregnated silica gel cartridges with KI ozone scrubber cartridge
	Plume	HPLC	Time-integrated	DNPH impregnated silica gel cartridges with KI ozone scrubber cartridge
Sample temperature ^f	Plume tunnel	Thermocouple	Continuous	K-Type thermocouples; T-Type only on APEX-2 sampling probes
APEX-2 plume temperature	Plume	Thermocouples	Continuous	Multiple T-type thermocouples
APEX-2 plume velocity	Plume	Pitot tube	Continuous	Standard pitot tube plus differential pressure cell
APEX-2 meteorological parameters ^g	Background	Propeller anemometer & wind vane	Continuous	Vaisala MAWS weather station

^a Filter holder design per Federal Test Procedure (FTP) published in 40 Code of Federal Regulations (CFR), Part 86. “Blow off” are gas-phase semivolatile species that have been released from the particulate deposited on the primary filter by the air flow passing through the medium.

^b Aluminum foil substrates from the ELPI were not analyzed due to insufficient mass.

^c These measurements were redundant and the data were not used.

^d The aethalometer measures “black” carbon which approximates elemental carbon content as determined from diesel engine testing at West Virginia University (Kinsey et al., 2006).

^e Post-test analysis of time-integrated Tedlar bag sample collected over the entire test period.

^f Temperature was not monitored in sampling lines.

^g Meteorological data provided by collaborators during APEX-1 and -3.

^h Data from these instruments are generally of low reliability and thus not reported in this paper.

CPC = Condensation Particle Counter

DMA = Differential Mobility Analyzer

DNPH = 2,4-Dinitrophenylhydrazine

EEPS = Engine Exhaust Particle Sizer

ELPI = Electrical Low Pressure Impactor

FID = Flame Ionization Detector

GC/MS = Gas Chromatography/ Mass Spectrometry

HPLC = High Performance Liquid Chromatography

IR = Infrared

NMOC = Nonmethane Organic Compound

NIOSH = National Institute for Occupational Safety and Health

PAH = Polycyclic Aromatic Hydrocarbon

PUF = Polyurethane Foam

QCM = Quartz Crystal Microbalance

SMPS = Scanning Mobility Particle Sizer

TEOM = Tapered Element Oscillating Microbalance

UV = Ultraviolet

XRF = X-ray Fluorescence

Table S-2.

Engines Tested in APEX-1, -2 and -3

Engine Model ^a	Airframe	Bypass Ratio ^b	Rated Thrust ^b (kN)	ICAO Smoke Number ^c				Test Campaign/Test Number
				T/O	C/O	App	Idle	
CFMI CFM56-2C1	Boeing DC-8	6	97.86	6.0	3.0	2.6	2.2	APEX 1 / All Tests
CFMI CFM56-7B24	Boeing 737-700	5.2	107.7	12.6 ^d	NA	NA	NA	APEX 2 / Test # 1 and 4
CFMI CFM56-3B1	Boeing 737-300	5.1	89.41	4.0	2.5	2.5	2.2	APEX 2 / Test # 2; APEX 3 / Test # 1 and 11
CFMI CFM56-3B2	Boeing 737-300	5.1	98.30	6.0	3.0	2.5	2.2	APEX 2 / Test # 3
General Electric CJ610-8ATJ (Turbojet) Starboard	Lear Model 25	na	13.12	NA	NA	NA	NA	APEX 3 / Test # 2 and 5
Rolls Royce AE3007A1E Starboard ^e	Embraer ERJ145	4.8	33.70	1.0	0	0	0	APEX 3 / Test # 3 and 4
Pratt & Whitney 4158 Starboard	Airbus A300	4.6	258.0	8.1 ^d	NA	NA	NA	APEX 3 / Test # 6 & Test #7
Rolls Royce RB211-535E4B Starboard ^e	Boeing 757-324	4.1	191.7	7.3 ^d	NA	NA	NA	APEX 3 / Test # 8 and 9
Rolls Royce AE3007A1/1 Starboard ^e	Embraer ERJ145	4.8	34.74	1.0	0	0	0	APEX 3 / Test # 10

^a All engines are turbofan except as noted.^b Civil Turbojet/Turbofan Specifications <http://www.jet-engine.net/civtfspec.html> or International Civil Aviation Organization (ICAO) Databank Issue 15-C.^c New engine certification data taken from ICAO Engine Emissions Databank Issue 15-C. T/O = take-off; C/O = climb-out; App = approach; NA = not available.^d Maximum SN; no power specified.^e These are internally mixed-flow turbofan engines.

Table S-3.

Summary of Fuel Flow Rates Measured at Different Engine Power Levels for Each Test Conducted

APEX	Test No. ^a	Aircraft	Engine	Fuel Type	Engine Conditions	Fuel Flow (kg h ⁻¹) at Percent Rated Thrust ^c														
						4	5.5	7	8.4	15	30	40	45	60	65	70	76	80	85	100
1	EPA-1	DC-8	CFM56-2C1	Base				436			992								2819	2969
	EPA-2							425			1023								2860	3181
	NASA-1					350	386	427		560	1012	1252			1998				2406	2906
	NASA-1a			High sulfur		336								1922	2098	2252			2898	3127
	EPA-3							438			964						2424		2840	3116
	NASA-2					345	381	413		543	955	1235		1855	2046	2191			2727	2984
	NASA-3					347	382	405		538	986	1255		1846	2053	220			2758	3051
	NASA-4			High aromatic		345	381	401		545	960	1220		1850	2023	2157			2708	2978
	NASA-5					345	395	410		545	989	1292		1930	2131	2247			2894	3176
2 ^d	T4	B737-700	CFM56-7B24	Fleet fuel	Cold ^b	336		418			1180	1544				2497			4131	
					Warm	313		381			1135	1498				2497			4086	
					Average	325		400			1158	1521				2497			4109	
	T2	B737-300	CFM56-3B1	Fleet fuel	Cold	341		422			1099	1403				2193			3528	
					Warm	345		418			1067	1367				2184			3559	
					Average	343		420			1083	1385				2188			3543	
	T3	B737-300	CFM56-3B2	Fleet fuel	Cold	372		440			1130	1444				2252			3677	
					Warm	368		422			1108	1412				2261			3650	
					Average	370		431			1119	1428				2256			3664	
	T1	B737-300	CFM56-3B1	Fleet fuel	Cold	300		397		654	1136		1618			2260			2903	3385
					Warm	300		397		654	1136		1618			2260			2903	3385
					Average	300		397		654	1136		1618			2260			2903	3385
3	T11	B737-300	CFM56-3B1	Fleet fuel	Cold	381		431		622	1090		1530			2179			2815	3564
					Warm			182		304	452		568			760			999	1226
					Average			182		304	453		568			762			999	1226
	T5	NASA Lear Model 25	CJ610-8ATJ (turbojet)	Fleet fuel	Cold			227		303	452		567			763			1009	1226
					Warm			227			452		567			763			1009	1226
					Average			227		303	452		567			763			1009	1226
	T3	Embraer ERJ 145	AE3007A1E	Fleet fuel	Cold				174	238	389		555			805			1082	1286
					Warm				173	235	392		563			810			1088	1299
					Average				173	237	391		559			807			1085	1293
	T4	Embraer ERJ 145	AE3007A1E	Fleet fuel	Cold				168	239	385		547			788			1050	1253
					Warm				167	231	384		549			786			1052	1252
					Average				167	235	385		548			787			1051	1252
	T10	Embraer ERJ 145	AE3007A1/1	Fleet fuel	Cold				179	233	372		524			750			971	1171
					Warm				178	231	371		529			767			982	1180
					Average				178	232	371		526			758			976	1175
	T6	A300	P&W 4158	Fleet fuel	Cold			610		1014	2245		3726		5658			7026		
					Warm			368		1097	2465		3834		5658			7026		
					Average			489		1056	2355		3780		5658			7026		

T7	A300	P&W 4158	Fleet fuel	Cold		600	1035	2230	3688	5702	7100	
				Warm		596		2252		5711	7200	
				Average		598	1035	2241	3688	5706	7150	
T8	B757	RB211-535E4-B	Fleet fuel	Cold	566	770	1191	2109	3178	4750		6096
				Warm	437	654	1178	2131	3436	4691		6449
				Average	501	712	1185	2120	3307	4720		6273
T9	B757	RB211-535E4-B	Fleet fuel	Cold	421	690	1221	2004	3068	4479		6233 6966
				Warm	506	668	1173	2037	3111	4551		6307 6987
				Average	464	679	1197	2021	3090	4515		6270 6976

^aNote that bleed air was extracted from the engine during tests EPA-1,-2, and -3 in APEX-1. No fuel flows were recorded by the airline during T1 of APEX-2.

^b“Cold” refers to increasing power in a stepwise fashion, and “Warm” indicates reducing power in a stepwise fashion.

^cFuel flows provided by aircraft operator.

^dFuel flows for APEX-2, T1, were derived from the same power conditions for T4, since no data were provided by the aircraft operator for this test.

5. Relationship of PM Mass Emission Indices to Fuel Flow

As discussed in the main text, a similar relationship between the PM mass emission index (EI_m) and increasing fuel flow was observed for all turbofan engines tested during the three APEX sampling campaigns. This observation is illustrated in Figure S-2 for a CFM56-7B24 turbofan engine tested in APEX-2 as compared to the CJ610-8ATJ turbojet engine sampled in APEX-3. As can be seen in this figure, a characteristic U-shaped curve of EI_m vs. fuel flow was determined for the CFM56-7B24, whereas for the CJ610-8ATJ turbojet, the EI_m increases linearly with increasing fuel flow.

5. Example Particle Size Distributions (PSDs)

All of the PSDs determined in the three APEX sampling campaigns generally contained a single primary mode for most engines and power conditions which was lognormally distributed. However, an accumulation mode was also observed at higher engine fuel flows (powers). The magnitude of this accumulation mode varied from engine to engine as illustrated in Figure S-3 for a CFM56-7B24 engine in APEX-2, and a CFM56-3B1 engine in APEX-3. As shown in the figure, the -7B24 model engine had relatively few accumulation mode particles in the PSD whereas the -3B1 model engine exhibited a very distinct accumulation mode at higher fuel flow.

6. Supporting Information Reference

Kinsey, J. S., Mitchell, W. A., Squier, W. C., Linna, K., King, F. G., Logan, R., Dong, Y., Thompson, G. J., Clark, N. N., 2006. Evaluation of methods for the determination of diesel-generated fine particulate matter: Physical characterization results. *Journal of Aerosol Science*, 37, 63-87.

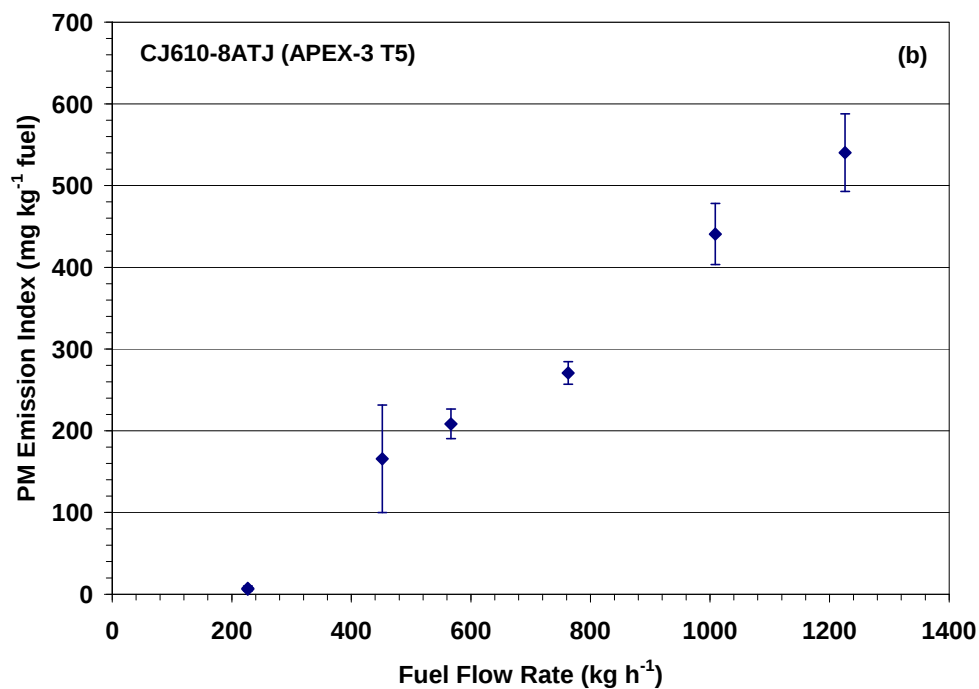
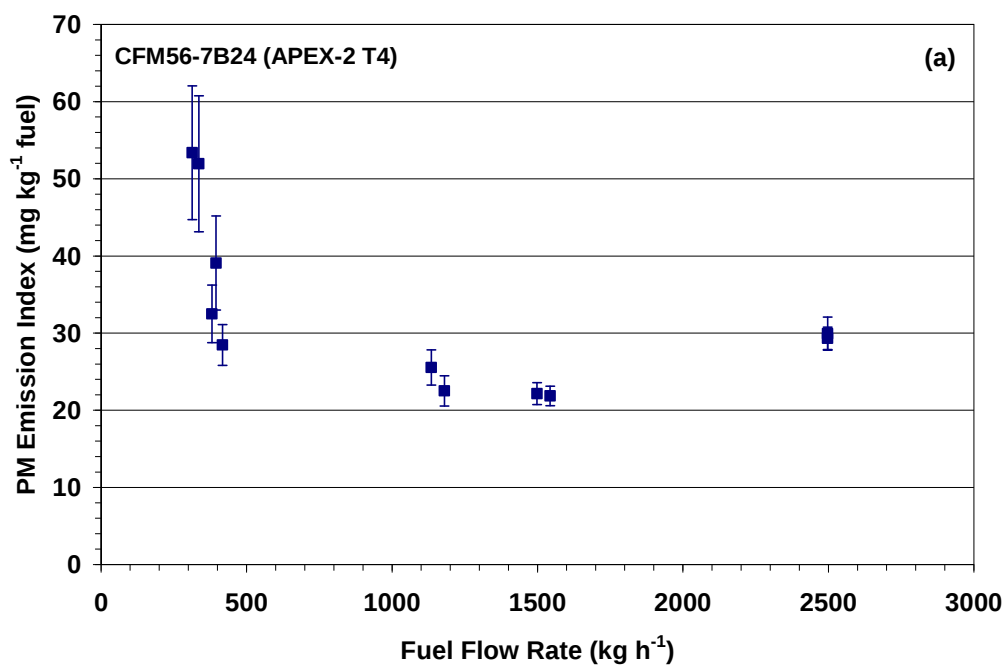


Fig. S-2. PM mass emission index as a function of fuel flow as determined by the nano-SMPS for a: (a) CFM56-7B24 turbofan engine in Test 4 of APEX-2; and (b) CJ610-8ATJ turbojet engine in Test 5 of APEX-3.

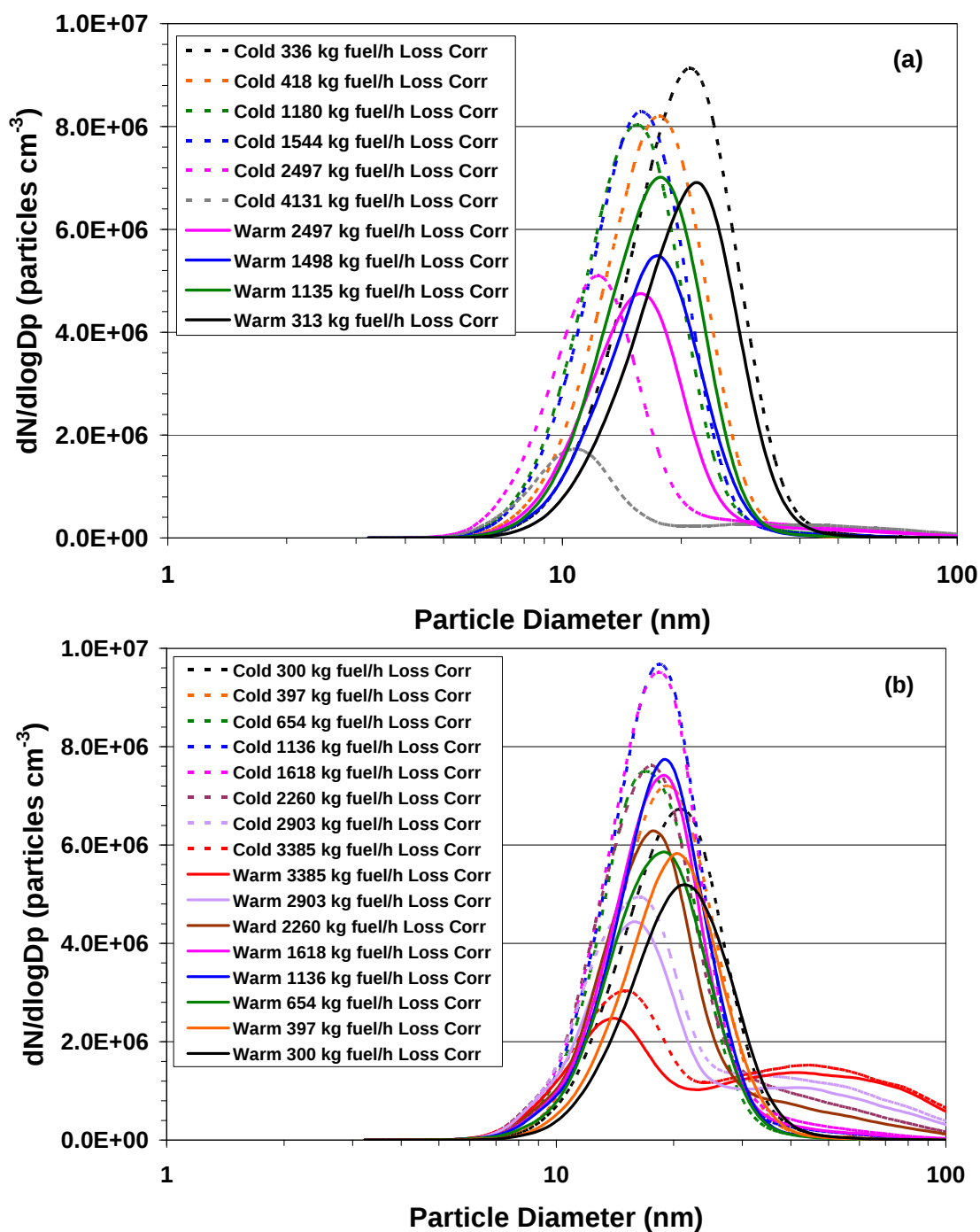


Fig. S-3. Average particle size distributions by fuel flow and engine temperature as measured by the nano-SMPS for: (a) a CFM56-7B24 engine in Test 1 of APEX-2; and (b) a CFM56-3B1 engine in Test 1 of APEX-3. Note the prominent accumulation mode present at high fuel flow for the -3B1 model engine.